

Airfoil surface		NACA0015
X(mm)		Y(mm)
%	X	Y(mm)
100	15.75	0.158
95	14.96	1.008
90	14.18	1.81
80	12.6	3.279
70	11	4.58
60	9.45	5.704
50	7.87	6.617
40	6.3	7.254
30	4.75	7.502
25	3.94	7.427
20	3.15	7.172
15	2.36	6.682
10	1.58	5.853
7.5	1.18	5.25
5	0.79	4.443
2.5	0.39	3.268
1.25	0.2	2.367
0	0	0
1.25	0.2	-2.367
2.5	0.39	-3.268
5	0.79	-4.443
7.5	1.18	-5.25
10	1.58	-5.853
15	2.36	-6.682
20	3.15	-7.172
25	3.94	-7.427
30	4.75	-7.502
40	6.3	-7.254
50	7.87	-6.617
60	9.45	-5.704
70	11	-4.58
80	12.6	-3.279
90	14.18	-1.81
95	14.96	-1.008
100	15.75	-0.158